



### The Challenge of Verification and Validation of Automated Planning Systems

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#### The Takeaway

- Verification and Validation of space mission planning systems is a challenging problem with several unique features:
  - Planning performed at a high level of abstraction (compared to spacecraft behavior).
  - V&V of model, not just code.
  - Availability of a simulation as an 'oracle'.
- V&V of mission planning can be automated by formalizing the relationship between the plan (actions, states) and the spacecraft behavior (commands and data).











- Mission Planning Systems are a key part of the ground segment of every space mission.
  - Mission planners generate courses of action to satisfy mission goals while respecting constraints.
- Like other software, the Mission Planning System (henceforth, the *planner*) must be verified and validated.
  - Verify that plans generated by the planner satisfy constraints and achieve goals.
  - Validate the constraints the planner enforces.





- Like (many) other applications, spacecraft mission planning places a premium on planning due to the high costs of defects and the complexity of the application.
  - Opportunities to test the planner prior to operations are limited to hardware simulations and ground-based testbeds.





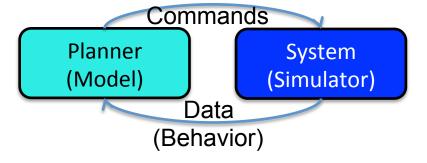
- Unlike other software, model-based planning systems change the nature of the software verification and validation problem.
  - The model is an input to the planning system.
  - Model verification and validation is different than software verification and validation!







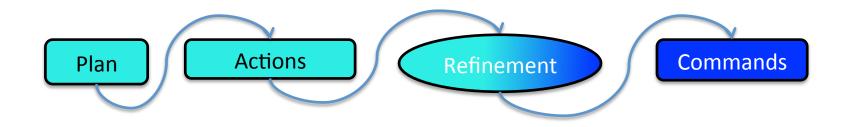
 Unlike other applications, the existence of a simulation offers an opportunity to validate the planner.







- Unlike other applications, abstraction is a key element of the V&V problem.
  - Planners approximate or abstract the actual spacecraft behavior.







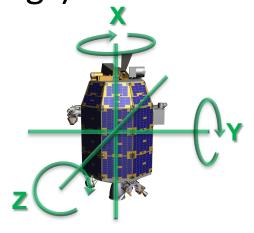
- Suppose we are developing a mission planning system for a spacecraft.
- How does a spacecraft change the direction (attitude) it is pointing?
  - A change of pointing is called a slew.





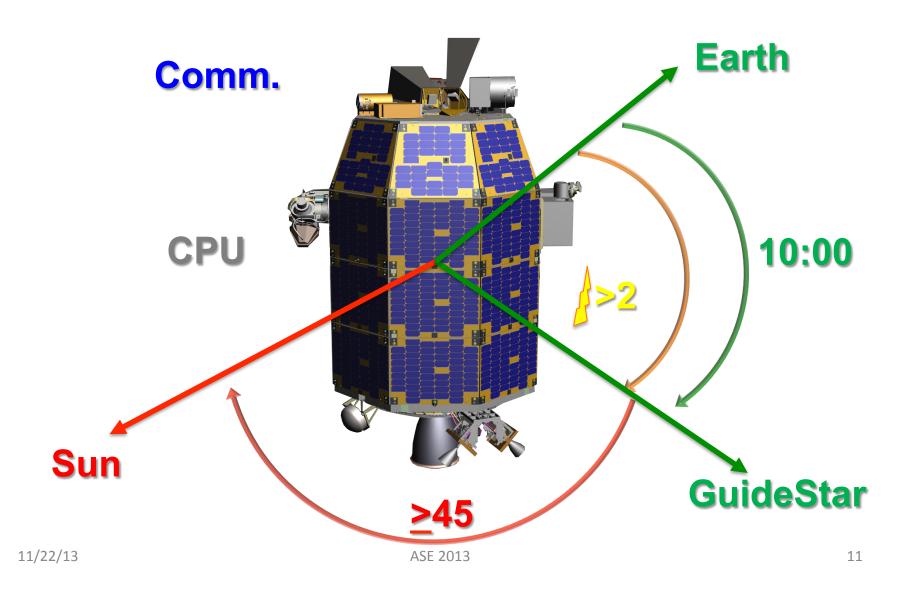


- Attitude is angles <x, y, z> around the x,y, and z spacecraft body axes.
- Slews are constrained by solar panel power generation, thermal, communications to Earth, sensor and instrument performance and safety (among other things).













### Mission Planning System Ground Network



Attitude Determination

Flight Software and Hardware

**Science Operations** 



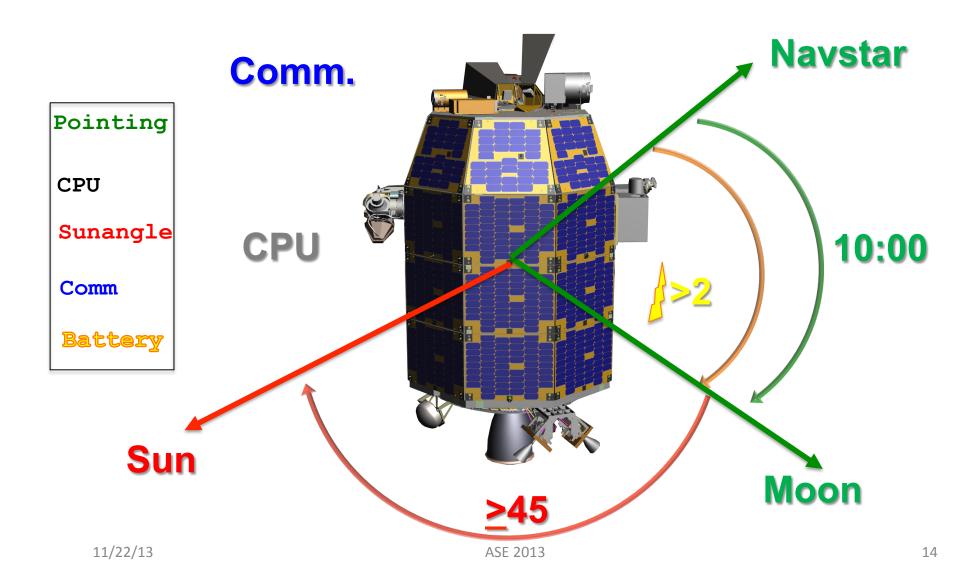


#### Mission Planning System

- A planning *model* consists of:
  - Objects things in the world.
  - States properties of things.
  - Actions ways of changing the properties of things.
- A planning problem consists of:
  - A model.
  - An initial state description.
  - A set of goal states.
- The planner reads the model, initial states, and goals, and produces a plan.
- There are myriad planning algorithms with many different properties.
  - Run-time, types of plans produced, etc.









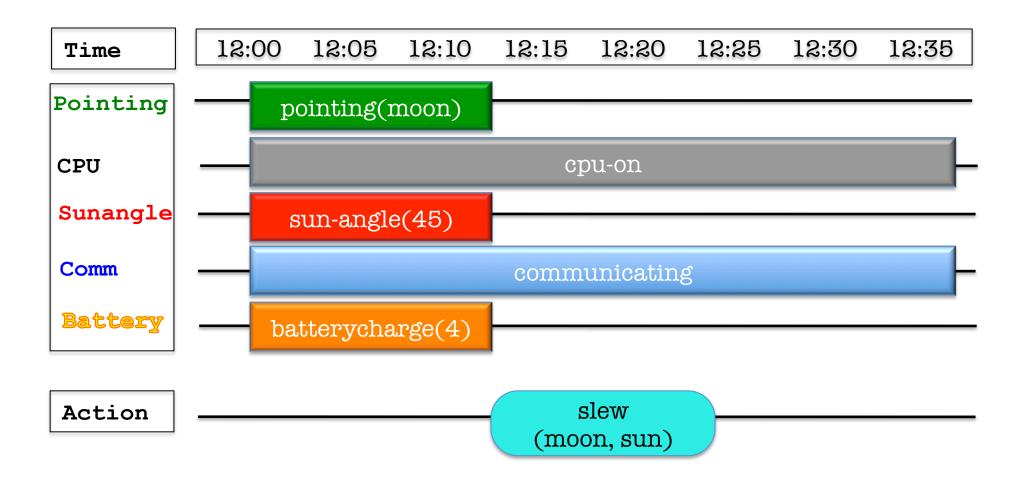
CPU



```
Action
       (:durative-action slew
         :parameters (?from - attitude
                       ?to - attitude)
                                                    Object
         :duration (= ?duration 10)
         :condition (and
Pointing
             (at start (pointing ?from))
                                                   Condition
             (at start (cpu on))
            (over all (cpu on))
                                                     States
            (at start (\geq = (sunangle) 45.0)
Sunangle
            (over all (>= (sunangle) 45.0))
            (at start (communicating))
Comm
            (over all (communicating))
Battery
            (at start (>= (batterycharge) 2.0)))
         :effect-
                                                   Effect
         (and
        (at start (decrease (batterycharge) 2.0))
             (at start (not (pointing ?from)))
             (at end (pointing ?to))))
                                                     State
```

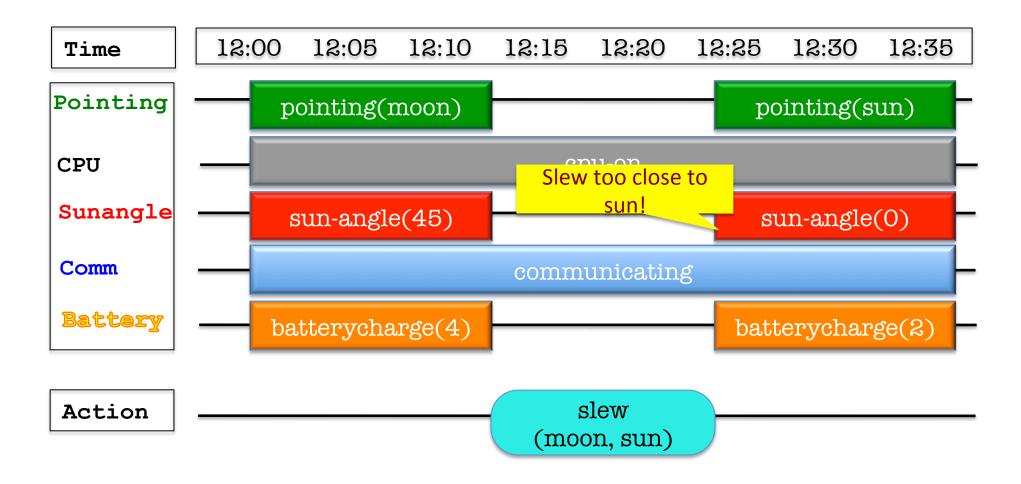






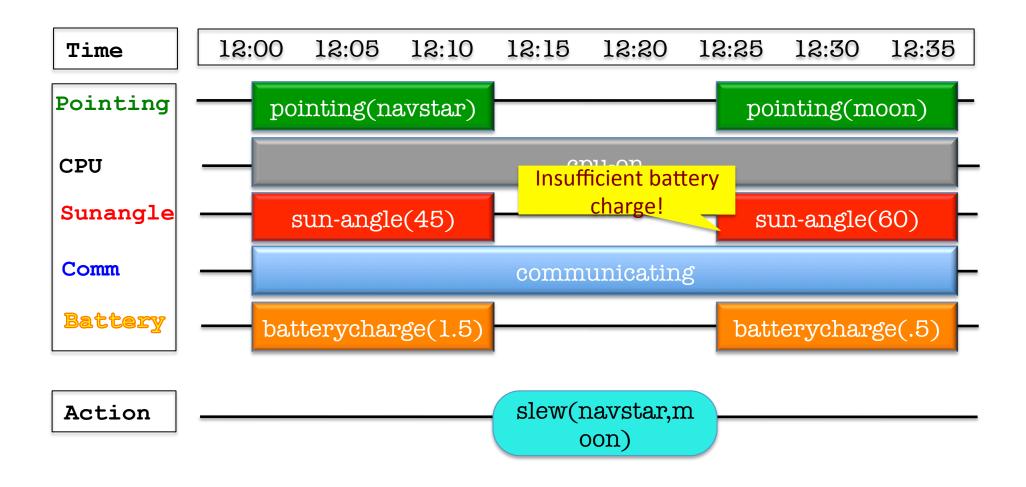






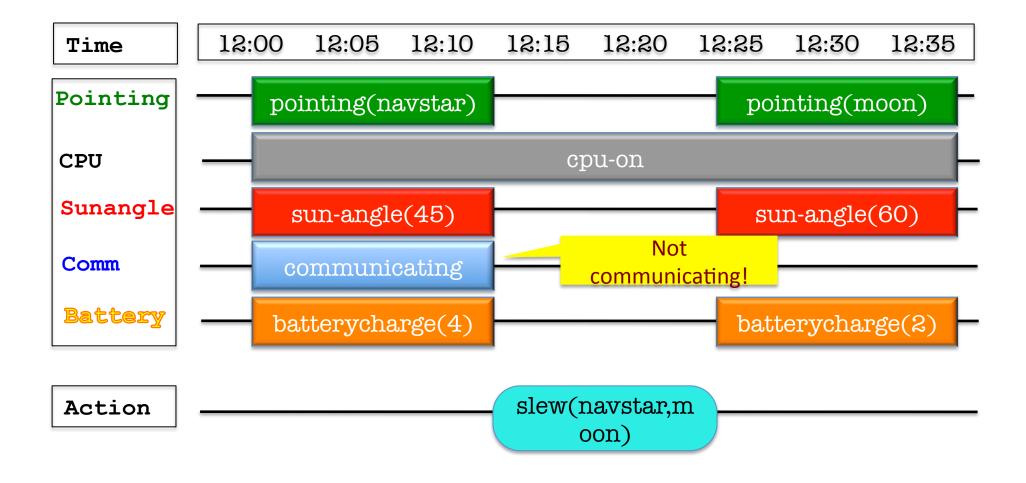






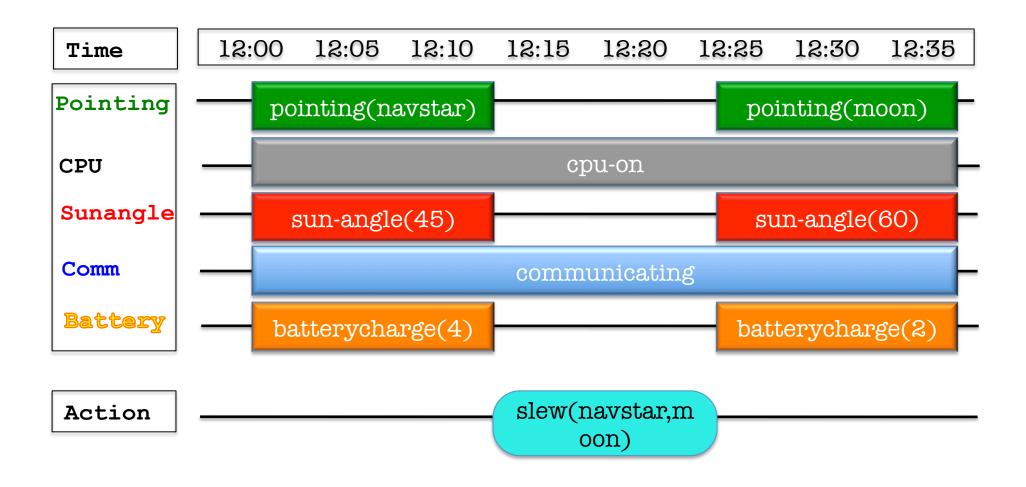








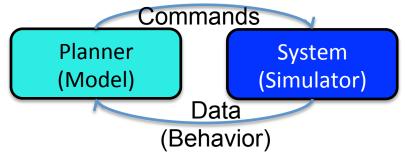








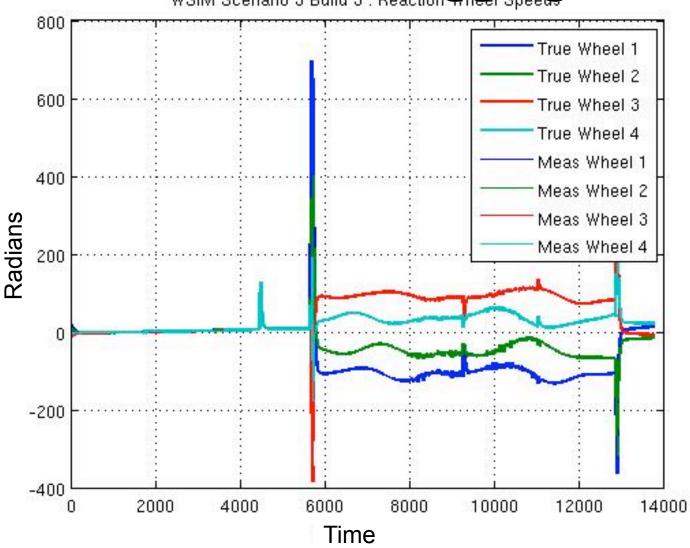
- Spacecraft simulation:
  - Accepts a sequence of commands
  - Coupled with an initial spacecraft state
  - Produces output corresponding to spacecraft behavior
  - Using software, hardware, or both to simulate the spacecraft







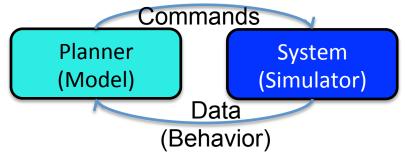
### Model Debugging WSIM Scenario 3 Build 5 : Reaction Research







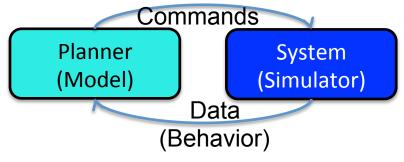
- What does it mean to validate the planner model with respect to the simulation?
  - Valid plans should execute on the simulator.
  - States predicted by planner should match (up to specified tolerances) the states produced by the simulator







- What does it mean to validate the planner model with respect to the simulation?
  - Does the slew take the expected amount of time?
  - Does the planned slew take place?
  - Does spacecraft point to proper place?







12:00

12:05

12:10

slew (navstar,moon)

Select RCS Pulsing Enable RCS
Thruster

Fire RCS
Thruster

X Body Rate < cutoff

Pointing Attitude

Y Body Rate < cutoff

Moon

Z Body Rate < cutoff





12:00 12:10

slew (navstar,moon)

Select RCS Pulsing Enable RCS Thruster Fir RCS Thrust r

X Body Rate < cutoff

Y Body Rate < cutoff

Z Body Rate < cutoff

Pointing Attitude

Moon





12:05

slew (navstar,moon)

ASE 2013

Select RCS Pulsing Fire RCS
Thruster
Enable RCS
Thruster

X Body Rate < cutoff

Y Body Rate < cutoff

Z Body Rate < cutoff

Pointing Attitude

Moon





12:00

12:05

12:10

slew (navstar,moon)

Select RCS Pulsing Enable RCS
Thruster

Fire RCS
Thruster

X Body Rate < cutoff

Pointing Attitude



Y Body Rate < cutoff

Z Body Rate < cutoff







12:00 | 12:10

slew (navstar,moon)

Select RCS Pulsing Enable RCS
Thruster

Fire RCS
Thruster

X Body Rate < cutoff

Y Body Rate < cutoff

Z Body Rate < cutoff



Sun



# Previous Work (Applications)



- Remote Agent [1], EO-1 [2]
  - Extensive model reviews.
  - Safety reviews to elicit potential hazards.
  - Automated tests stochastically generated by perturbations of nominal scenarios.
  - Executed on simulation platforms of varying fidelity where spacecraft, operations, and safety constraints were checked.





# Previous Work (Academia)



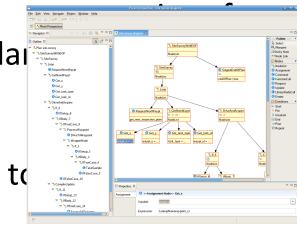
- itSimple [3]
  - Allows some domain behavior modeling using UML object diagrams.
  - Generated plans can be checked against the UML.
- KEEN [4]
  - Similar to itSimple, but uses Timed Game Automata (TGA) instead of UML as domain model.
  - Emphasis on temporal planning domains and temporally flexible plans.
- PDVer [5]
  - Plan domain properties specified in LTL (Linear Temporal Logic).
  - Specification of test cases (goals) automatically from LTL.



# Previous Work (Academia)



- VAL [6]
  - Given a plan and a model, determines whether the plan satisfies the constraints in the domain.
  - Limited ability to automatically fix plans.
- Model checking as plan verification [7]
  - Employs Java Pathfinder to check
     PLEXIL, a language and plansystem.
  - Requires a system model (or and set of properties





# Previous Work (Summary)



- There are tools to assist in verification of plans against planning models.
- There are tools to assist in test case generation and model verification.
- Few to no tools to assist in validation of models.
- No tools to assist in validation against simulations.



## Integrated Model Development Environment

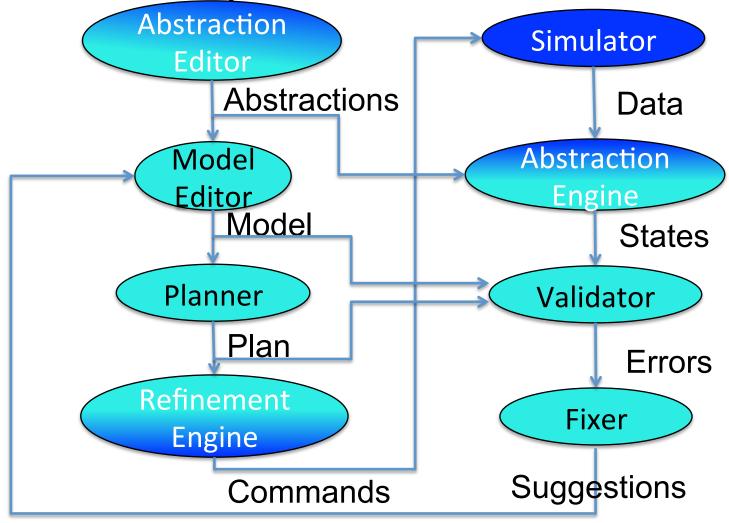


- The Integrated Model Development Envrionment (IMDE) bridges the gap between planning and simulation:
  - Modeler documents translation of plans to command sequences.
  - Facilitates automatic generation of simulations.
  - Modeler documents translation of spacecraft data to plan states.
  - Facilitates automatic translation of simulation output into planning model states.
  - Enables comparison of planned and simulated spacecraft behavior.
- See [8,9] for a more complete description.



# Integrated Model Development Environment







### IMDE Architecture: Abstraction Editor



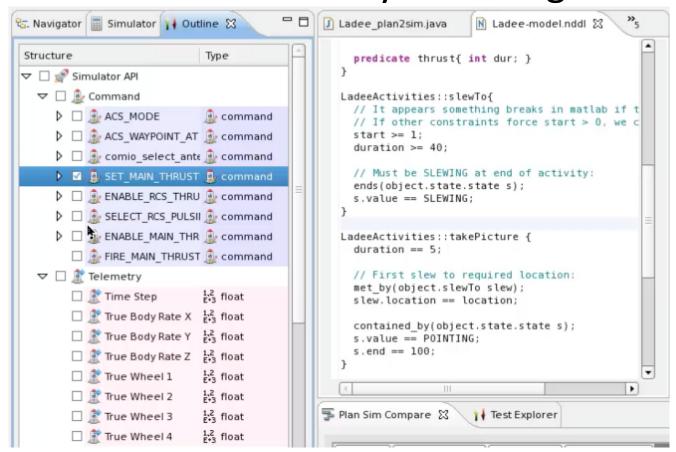
- Document relationship between model (actions, states, parameters, values) and spacecraft (simulation) command and data specification.
- Provides human-readable traceability between systems.
- Formal representation of abstractions is input to Refinement/Abstraction engine.



# IMDE Architecture: Abstraction Editor



Commands and Telemetry Browsing:

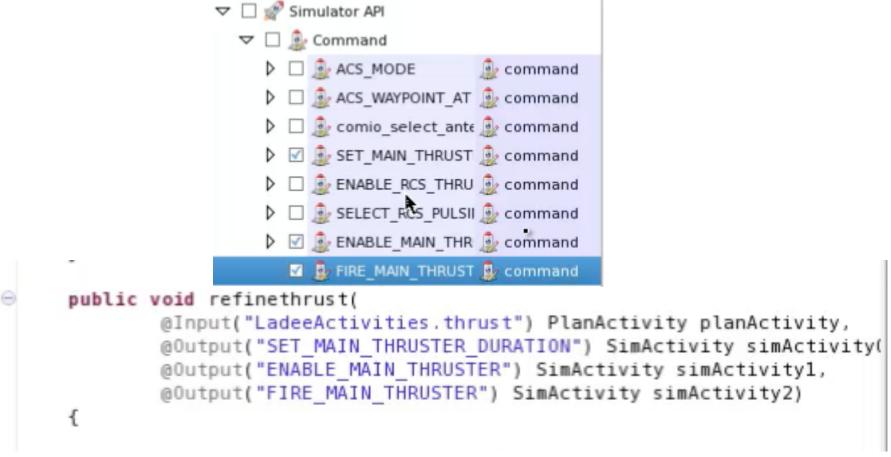




## IMDE Architecture: Abstraction Editor



Refining Planner actions into commands:





### IMDE Architecture: Abstraction Editor



Abstracting spacecraft telemetry into states:

```
public boolean abstractBodyNate(
       @Input("True Body Rate X") String vall,
       @Input("True Body Rate Y") String val2.
       @Input("True Body Rate Z") String val3,
        List <DSAParameter> params)
   double dl = Math.abs(Double.parseDouble(vall));
   double d2 = Math.abs(Double.parseDouble(val2));
    double d3 = Math.abs(Double.parseDouble(val3));
   double cutoff = 0.001;
   // d3 settles to -0.001, not 0
    boolean moving = (dl > cutoff || d2 > cutoff); // || d3 > cutoff)
   boolean val = (dl > cutoff || d2 > cutoff || d3 > cutoff);
    params.add(new DSAParameter("value", "" + (moving ? "SLEWING" : "P
    return true:
```



### IMDE Architecture: Abstraction Refinement Engine



#### • Refinement:

- Transforms plan and initial state into simulation input and command sequence.
- Transformation defined by 'inverting' abstractions created in Abstraction Editor.

#### Abstraction:

- Transforms simulation output into 'predicted' plan states.
- Transformation uses abstractions created in Abstraction Editor (for data to states only).



### IMDE Architecture: Validator



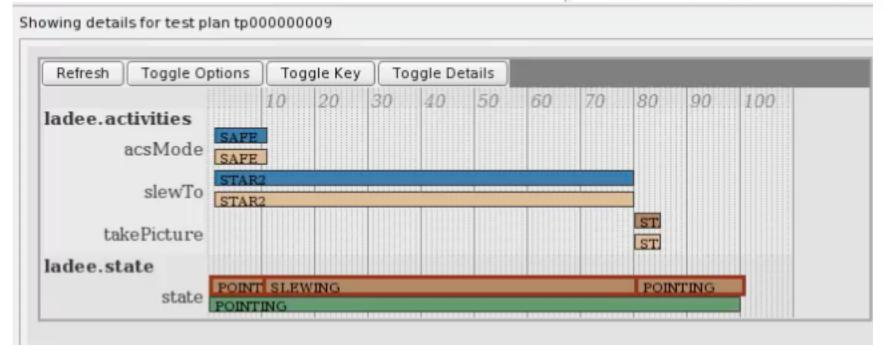
- Compare states as generated by planner to those abstracted from simulation.
- Differences indicate a discrepancy between plan model and simulation behavior.
- Can also report simulator errors, constraint violations not caught by simulator, etc.



# IMDE Architecture: Validator



✓ ① Errors (3 items)
 ② OPEN\_CONDITION: LadeeState.state
 ② OPEN\_CONDITION: LadeeState.state
 ③ OPEN\_CONDITION: LadeeState.state
 ✓ ▲ Warnings (1 items)
 ▲ DISCREPANCY: LadeeState timeline





# IMDE Architecture: Validator



× tp000000003	NA	NA	Yes	40	LadeeActivities. location	STAR2	10	LadeeActivil mode	FINE_POINTING
× tp000000007	NA	NA	Yes	40	LadeeActivities. location	STAR2	1	LadeeActivit mode	SAFE
× tp000000011	NA	NA	Yes	40	LadeeActivities. location	STAR2	10	LadeeActivit mode	SAFE
× tp000000015	NA	NA	Yes	40	LadeeActivities. location	STAR2	1	LadeeActivit mode	FINE_POINTING
X tp000000000	NA	NA	Yes	60	LadeeActivities. location	STAR2	1	LadeeActivit mode	FINE_POINTING
X tp000000004	NA	NA	Yes	60	LadeeActivities. location	STAR2	10	LadeeActivit mode	FINE_POINTING
(i) tp000000008	3	0	Yes	60	LadeeActivities. location	STAR2	1	LadeeActivil mode	SAFE
① tp000000012	3	0	Yes	60	LadeeActivities. location	STAR2	10	LadeeActivit mode	SAFE
💢 tp000000001	NA	NA	Yes	80	LadeeActivities. location	STAR2	1	LadeeActivil mode	FINE_POINTING
💢 tp000000005	NA	NA	Yes	80	LadeeActivities. location	STAR2	10	LadeeActivit mode	FINE_POINTING
<b>t</b> p000000009	0	0	Yes	80	LadeeActivities. location	STAR2	1	LadeeActivi1 mode	SAFE
① tp000000013	3	0	Yes	80	LadeeActivities. location	STAR2	10	LadeeActivit mode	SAFE
¥ tn0000000002	NA	NA	Yes	90	LadeeActivities location	STAR2	1	LadeeActivit mode	FINE POINTING



### IMDE Architecture: Fixer



- If a command in the slew fails to execute:
  - Planner model could be missing one or more conditions the simulator enforces (e.g. thermal).
  - Planner model conditions could be wrong.
  - Abstraction of commands to action could be wrong (e.g. sequence isn't right, duration wrong).



#### IMDE Architecture:



#### **Fixer**

- If there is a discrepancy between planned and simulated states, or the planned action is incompatible with the simulated states:
  - Planner model effects could be missing (e.g. thermal effects).
  - Planner model effects could be wrong.
  - Abstractions of telemetry to states could be wrong (e.g. body rate threshold for pointing could be wrong).



#### IMDE Architecture:



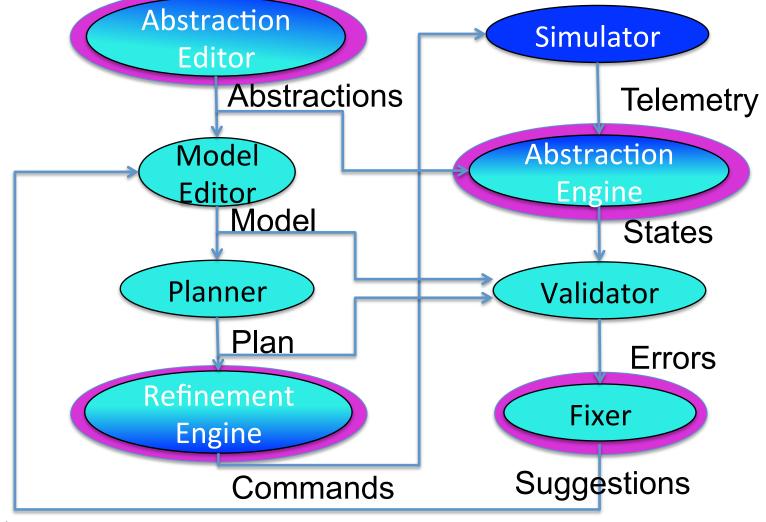
#### **Fixer**

- Given a discrepancy between planner model and simulator
  - Identify part of model that is to blame.
  - Ideally: Suggest changes to model that will address the problem.
  - Problem: the error could be almost anywhere!



# Integrated Model Development Environment









- All abstractions either
  - Map a domain of a telemetry variable into a domain of smaller cardinality
    - $f(X) \Rightarrow Y \text{ s.t. } |X| > |Y|$
    - Special cases: eliminate element of a discrete domain, map reals to integers, map integers to positive integers
  - Map n variables to m<n variables</li>
    - $f(x_1...x_n) \Rightarrow \{y_1...y_m\}$
    - Cardinality must still be reduced, e.g.  $|X_1| |X_2| ... |X_n| < |Y_1| |Y_2| ... |Y_m|$
    - Special cases: eliminate variable





- Planner / plans
  - N actions
  - $-P_n$  action conditions / effects for action n
  - S states
  - M state instances in a plan
- Simulator
  - C<sub>n</sub> commands in refinement of action n
  - T telemetry items
  - R values for each item per run
- Model
  - A abstractions
- We expect M << TR</li>

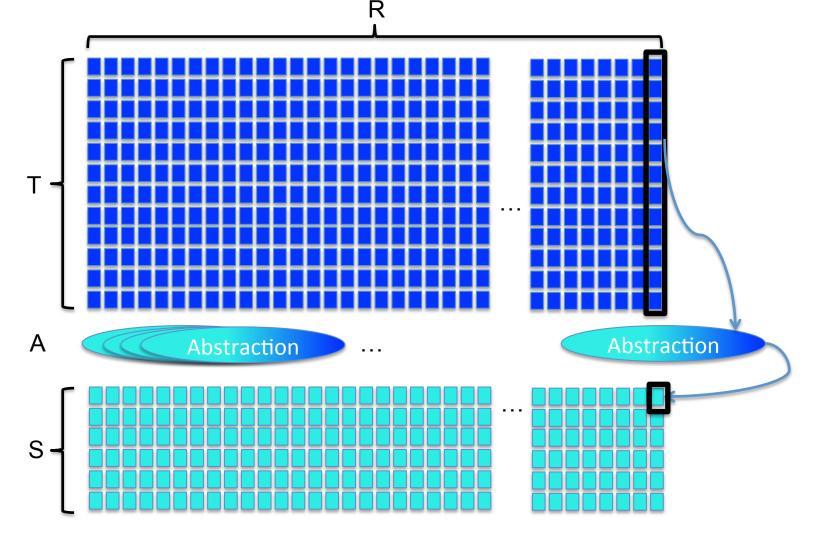




- Generate refinement from plan
  - $-\sum C_n$  (N actions,  $C_n$  commands in refinement of action n)
- Generate abstraction
  - TRA + 2SR (T telemetry items, R values for each item per run, A abstractions, S state types)
  - First pass is to generate states: for all R, for all abstractions A, each abstraction uses at most T telemetry items; writes at most S states holding at each instant R instants. This gets us runtime TRA+SR.
  - 2nd pass is to determine start / end times of states; this is another SR.
  - (There are important assumptions about the form of the abstractions i.e. they only use values at one time tic)







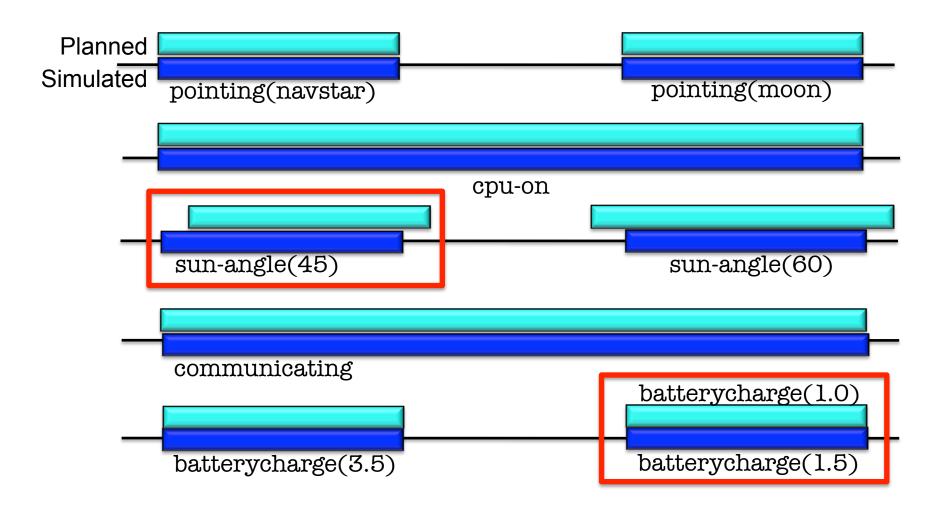




- Generated warnings
  - M<sup>2</sup> (M distinct state instances in plan, check to see if simulated start / end time match compared to planned start / end times; also check to see if simulated states internally consistent)
- Generate discrepancies
  - $-M(\Sigma P_n)$  (N actions,  $P_n$  action conditions / effects)
  - For each condition/effect of an action, may need to search the M states to match conditions / effects

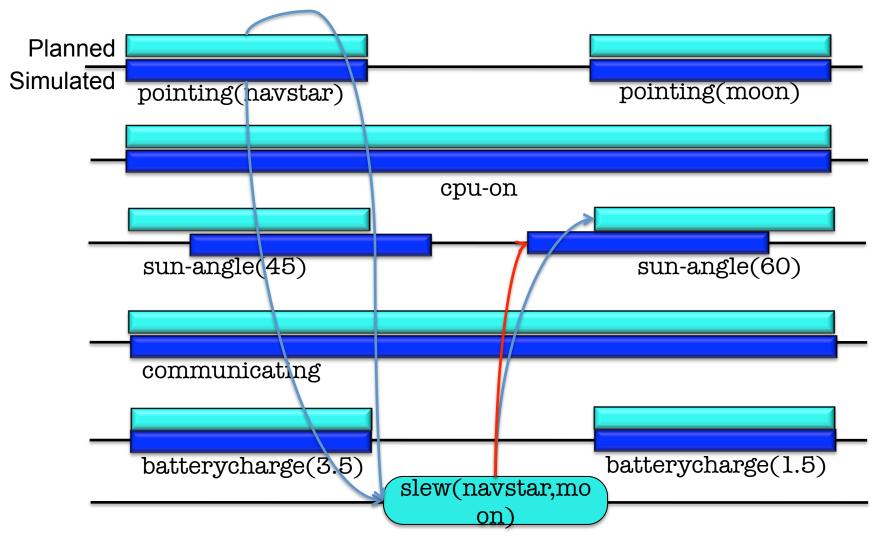










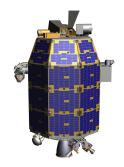






#### A Few Words About LADEE

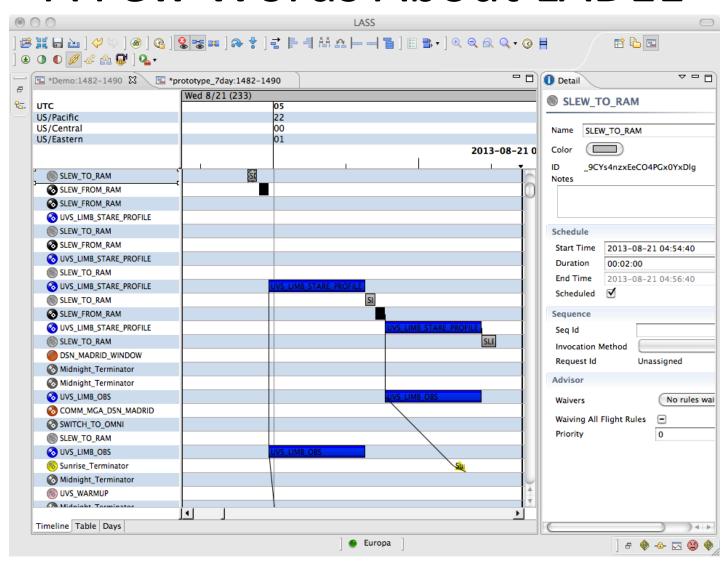
- How would this approach need to scale for LADEE?
  - ~600 Commands
  - ~25000 Telemetry / data
  - 122 Activities
  - 27 States
  - 21 Numerical Resources
- The LADEE planner model has ~ 12000 lines.
- Simulation data produced at 10Hz (cycles / second)







#### A Few Words About LADEE







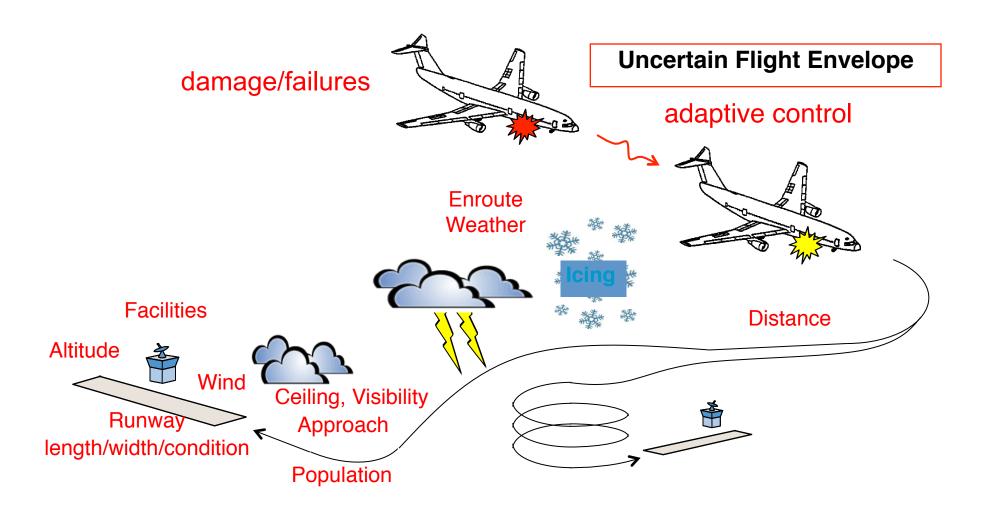
#### **Future Work**

- How can errors be identified and fixed for different modeling language features, such as uncertainty, parameter functions, and decompositions? (e.g. learning)
- How can the architecture be adapted to suggest changes for plan quality?
- How can we take advantage of white box simulators? auto-generate refinements to sim commands? auto-fill model? [13]
- See [14] for tools for authoring abstractions.





#### **Future Work**

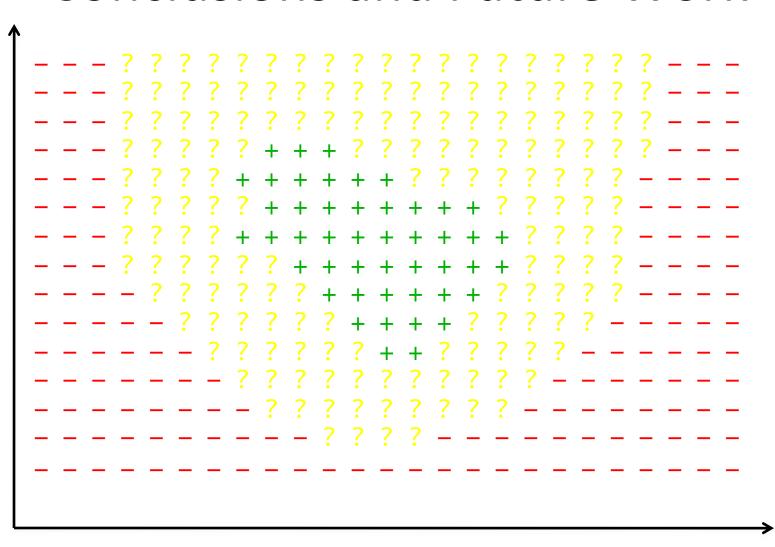




Airspeed



#### **Conclusions and Future Work**







#### **Future Work**

- Future mission planning systems could explicitly incorporate uncertainty:
  - (Limited) Contingency Planning [10]
  - Optimal policy generation for (Partially Observable)
     Markov Decision Processes ((PO)MDPs) [11]
  - Conformant planning [12]
  - 'Generate' using classical planning plus 'test' using simulation [15]
- This will make the V&V challenge harder.





#### The Takeaway

- Verification and Validating of planning models is a challenge.
- Extending the notion of the traditional software IDE as our solution:
  - Expose simulation API in plan model editor.
  - Providing 'good UI'.
  - Document relationship between elements (abstractions) enables automation:
    - Transform plans to simulation inputs.
    - Transform simulation outputs.
    - Comparison with plans to generate discrepancies and warnings.





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#### References

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- [14] Bell, S., Kortenkamp, D., and Zaientz, J. A Data Abstraction Architecture for Mission Operations. In *Proc. of the International Symposium on AI, Robotics, and Automation in Space*, 2010.
- [15] M. Fox, D. Long and D. Magazzeni (2012) "Plan-based Policies for Efficient Multiple Battery Load Management", Journal of Artificial Intelligence Research, Volume 44, pages 335-382





### Thank You!







#### **BACKUP SLIDES**





#### Outline

- Challenge
- Automated Planning Systems
  - Example
  - Mission Planning
  - Model-Based Planning
  - Model Debugging
- Previous and Related Work
- Integrated Model Development Environment (IMDE)
  - Concept of Operations
  - IMDE Architecture
  - IMDE Algorithms
  - A Few Words about LADEE
- Conclusion and Future Work



#### Example (Redux)



- Simulate a plan solution with a single slew to Earth and a goal (pointing Earth).
  - Construct simulation commands from plan
  - Construct initial state from plan
  - Select other initial conditions not part of plan initial state
- Simulation output shows no change in attitude.
  - Must be an error in the model or abstractions because the goal fails.
- No other output suggests anything went wrong.
  - No information to identify error!
- Simulate more slews with different initial states.
  - Some simulations show successful slews, and others show no slew.
  - The most noticeable pattern is that whenever the simulator cpu-on is zero, the slew is not executed.
- Fix the model:
  - add the cpu-on simulator state variable to planning model
  - add a cpu-on precondition to the slew activity OR
  - add a turnOnCpu() simulator command to the refinement OR
  - add the turnOnCpu() simulator command to the planning model and add a temporal constraint that slew be preceded by turnOnCpu().





#### Model-Based Planning

- Example: spacecraft attitude change (slew)
  - pointing a state variable indicating spacecraft orientation or change of orientation
    - Takes as parameter one of a (possibly infinite) set of directions, attitudes, or poses
  - communicating a state variable indicating whether or not the spacecraft is communicating with a ground antenna
  - batterycharge a state variable (resource) indicating battery state of charge
    - numerical value (hours of charge remaining)
  - sunangle a state variable tracking where the sun is relative to the spacecraft
    - Numerical value (vector from spacecraft to sun)



#### Motivation (redux)



- Model checking does not help validate model in system context.
  - Prior plan verification/validation work
    - verifies constraints/properties encoded in another language,
    - ensures against foreseeable problems,
    - verifies self-consistency, or
    - validates plan as a solution.
  - We are interested in checking against system behavior as represented by simulation.
  - Simulation can be used to check simultaneously for all problems, many of which
    - may not be expressible in the model checking language or
    - may be unforeseeable.



#### Motivation (redux)



- NOT auto-generating planning model from simulation, because
  - The planning domain modeler does not build the simulation.
  - The simulation is usually a black-box.
  - Even if it weren't it is too difficult to transform the white-box model into our planning representations.
  - And the simulator can't be used for up-front planning. (It is not designed to be used this way).



### Integrated Model Development 🕠



#### Environment

- Instead of testing all possible plans, start with a single model element.
- For example, simulate **slew** for different initial states and parameterizations.
- Then simulate all two-activity plans (slew and another activity) with different temporal relations.
- Continue while growing numbers of activities.
- May be able to avoid plans like turnOnCpu → turnOnCpu → slew...
- Likelihood of finding error decreases and confidence in model increases.
- Is there a point where we can stop and claim **slew** is correct/ valid?
  - No, but we may with assumptions of causal independence.
- If each action model is valid, the entire model is valid!
- Validate model as it is built from scratch!



#### **Technology Foundations**



- DAASA
  - S/C Telemetry abstraction authoring and execution
- VAL
  - Validation of plans against specifications
- ItSimple, PRIDE
  - Integrating modeling and simulation
- LOCM
  - Learning plan models from plan traces
- PAGODA
  - Learning plan models from simulations
- SLATE
  - Validating composite actions/behaviors through testing





## Questioins

- Finite modes and mode transitions in LADEE
   FSW; does this map to LASS? How?
- Put 'talk takeaway' on one slide (upfront? At end? Both?)
- How to emphasize future of IMDE?
- Didn't talk about mnultiple runs for either abstraction or uncertainty; how to cover that?





# Example

- Attitude Control System Flight Rule
- <u>Rule:</u> The final attitude of a turn shall place the Sun greater than 45 degrees away from the S/C Y axis (+Y/-Y)
- <u>Rationale:</u> Flight software will reject pointing commands that place the Sun within 45 degrees of the Y axis. Rejection of pointing command will result in the slew not being executed and the spacecraft remaining in its current ACS mode and attitude.
- Criticality: Category B
- Mission Phase Dependency: Cruise, Orbit, Science
- <u>Commands Affected:</u> AC\_SET\_DV\_RCS\_TGT, AC\_SET\_INERTIAL\_TGT, AC\_SET\_SCI\_NADIR\_TGT, AC\_SET\_SCI\_TRACK\_TGT
- <u>Cognizant Individual:</u> Attitude Control System Engineer
- <u>Notes:</u> Flight software will permit slewing through this keepout zone, planning should avoid slewing through this zone if possible.





# Model-Based Planning

- Model elements in more detail
  - Objects things in the world
    - E.g. targets, spacecraft components
  - States and Resources
    - Properties of things e.g. available power, mode of system
    - Properties change over time; the term State Variable refers to the history of the changing property
    - Resources are states represented by numbers; resources are depleted and produced by actions

#### Actions

- Conditions what must be true, when, and for how long, for an action to have the desired effect
- Effects what changes when the action is executed, when, and for how long





# Example

#### ACS

- 4 reaction wheels
  - only 3 used
  - Which ones depends on type of slew
- Reaction wheel commands
  - Start, change speed, off
- Reaction wheel data
  - RPM, direction, on, off
- Startracker
  - Sun position

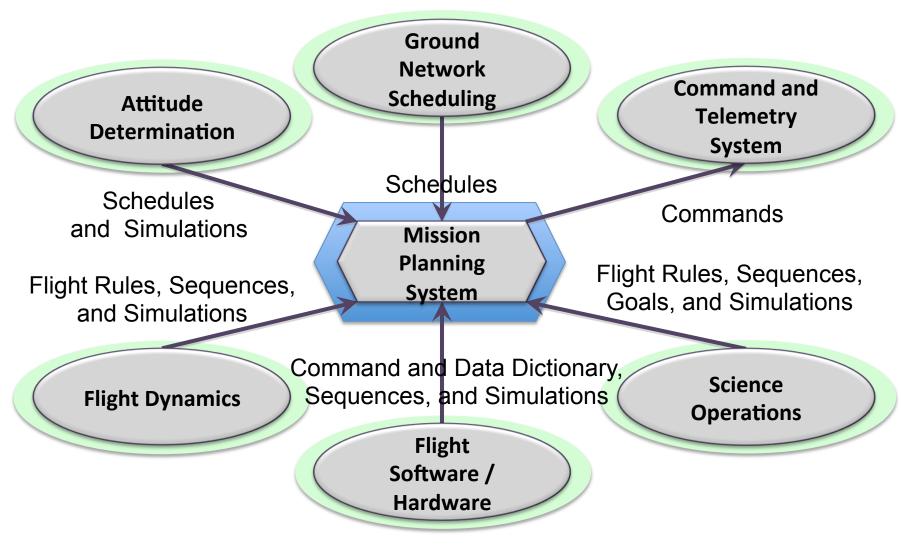
#### EPS

- Subsystems powered
- Battery state of charge
- Voltage on channel
- CPU
  - On, Off
- IMU
  - On, off
  - X,Y,Z attitude
  - X,Y,Z rate of change





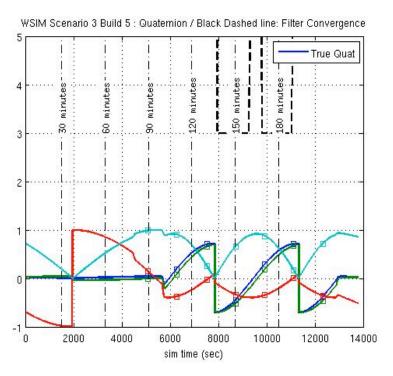
# Mission Planning System

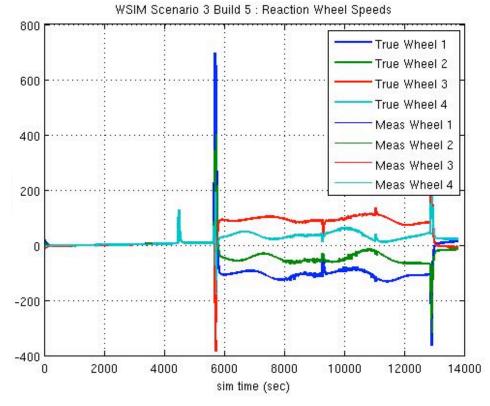






# Model Debugging



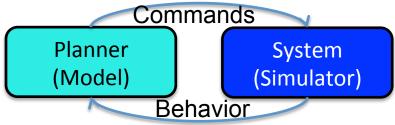






# Model Debugging

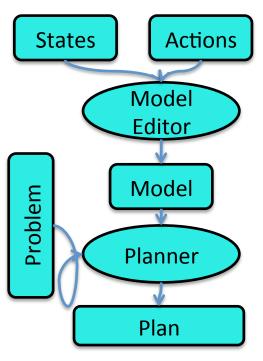
- Manually document system commands correspondence to activities
- 2. Write action and state/resource models using text editor based on available descriptions of system
- 3. Manually create planning problems and run planner
  - look at solutions to debug model
  - if no solution, figure out why
- Translate plans to commands; run commands through simulator
- 5. Translate simulator output out if it did what it expected
- 6. correct and augment model based on new knowledge





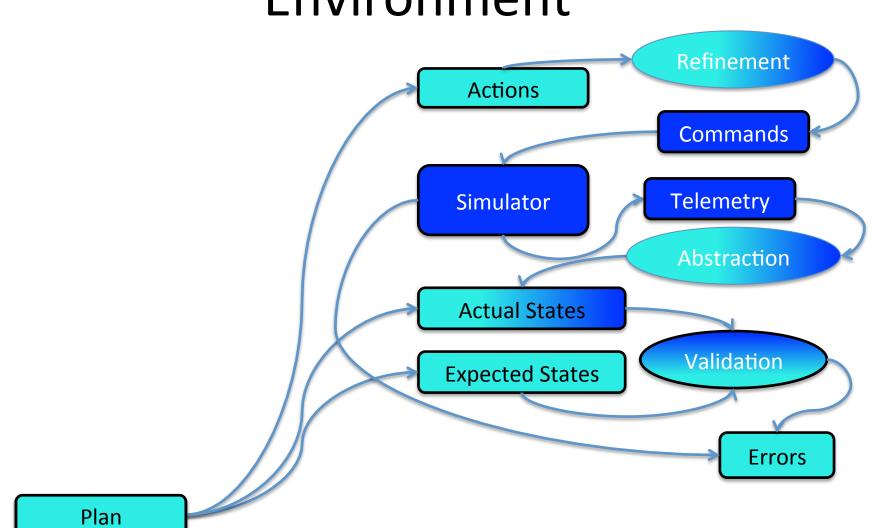
# Integrated Model Development 📌 **Environment**





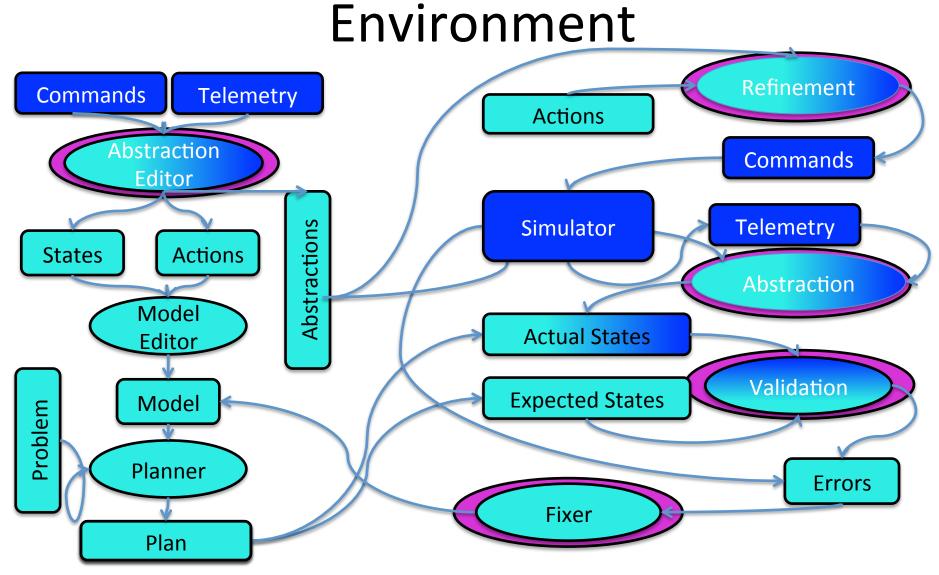


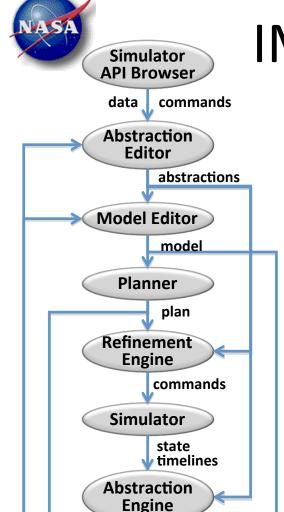
# Integrated Model Development 💝 **Environment**





# Integrated Model Development 📌





simulated execution

errors

Validator

**Fixer** 

suggestions

### **IMDE** Architecture



- Abstraction Editor
  - Store relationship between plan and simulation entities
- Model Editor
  - Integrated tightly with simulation
- Refinement Engine
  - Translations between plans and simulations; employs stored abstractions
- Abstraction Engine
  - Translates simulation traces to 'asexecuted' plans
- Validator
  - Checks 'as-executed plans' against model
- Fixer
  - Proposes fixes to model



# IMDE Architecture: Abstraction Editor

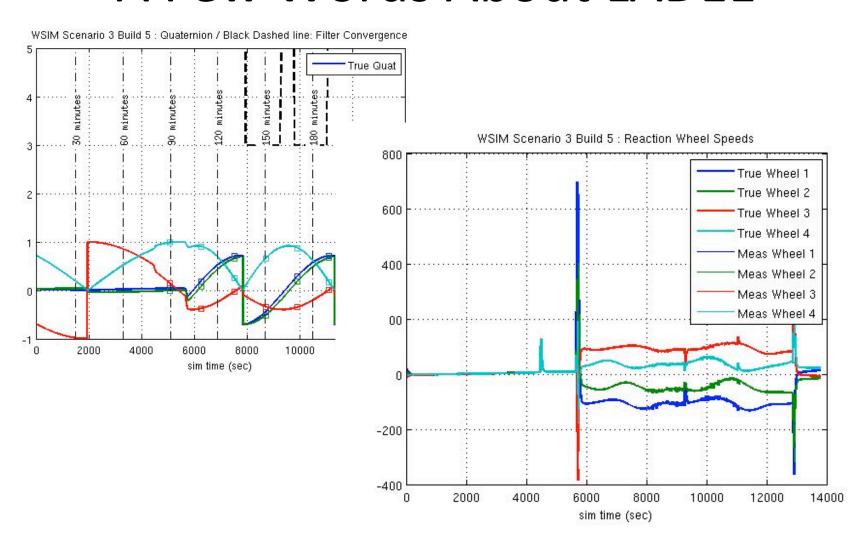


- There are some complexities in abstracting spacecraft telemetry into states
  - Different telemetry rates
    - Mostly spacecraft 'frame' all data but not always
  - State duration and transitions
    - If body rate exceeds cutoff for 1 millisecond does the state transition from pointing to slewing? (YES!)
    - Does this need to be declared in the abstraction? (YES!)





### A Few Words About LADEE







# Model Debugging

- What could go wrong?
  - Slew commands not executed.
  - Slew commands executed at wrong time.
  - Slew takes longer or shorter than expected.
  - Slew commands execute, but simulation output shows no change in attitude.
  - Slew commands executed but simulation shows spacecraft pointed wrong. (Or two directions at the same time!)
  - Slew executes when it is not supposed to (e.g. violates sun pointing constraint)
  - Slew works fine, but other simulated behavior is not as expected (e.g. excess power consumption)





# Conclusions and Future Work

- But now you have to V&V:
  - (Probability distributions over) sensor input values
  - (Probability distributions over) possible action outcomes
  - (Probability distributions over) unexpected events
  - (Probability distributions over) plan quality





## Conclusion and Future Work

- How can a complete but tractable space of test case plans be identified for activity model validation?
- Can a single test case contribute to the validation of multiple model elements?
- Where is the true cause of the error? model? command refinement? data abstraction? simulator? actual system?
- What are the features of a learning problem for classifying an error? numbers of actions of each type? temporal ordering? initial state? simulation state at time of error?
- How can suggested fixes be generated for these errors?





# Example

- When planning the slew:
  - How long does a slew take?
  - When is slewing allowed? Not allowed?
  - How does the abstract slew activity map to sequences of spacecraft commands?
  - How do we know when continuous <x, y, z> is pointing to discrete locations? (e.g. the Earth)



# Integrated Model Development 🔀 **Environment**



- The IMDE automates detection of errors in model based on inconsistencies between the plan and simulation.
- The IMDE automates identification of possible causes of errors from multiple runs.
  - Automate suggested fixes to model.
- More information in [8,9].